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NASA Overview/Update FAA ASCENT Meeting

April 2019

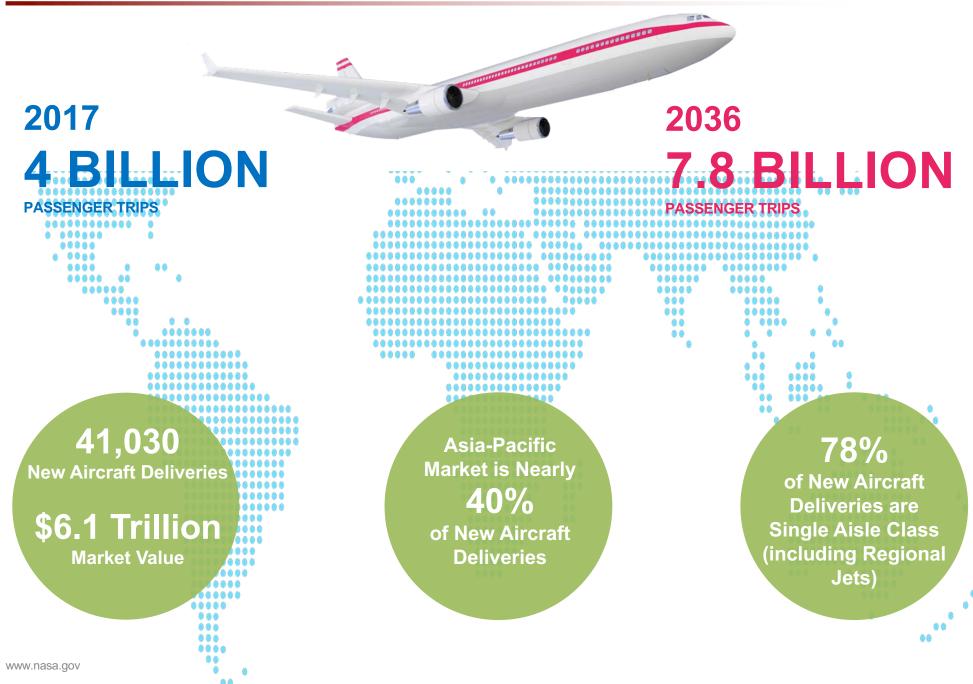
Angela Surgenor Technical Integration Manager, Advanced Air Vehicles Program NASA Aeronautics Research Mission Directorate **Brief Outline**



- Global Growth in Aviation
- Emerging Markets Integrated Challenges
- NASA Vision and Strategy
- FY2020 Budget request
- Subsonic Transport Technology Strategy
- Enabling U.S. Leadership in Subsonic Transport Markets
- Electrified Aircraft Propulsion Strategy
- Alternative Fuel Research NASA
- Landing/Takeoff Noise and Emissions Procedures for Supersonic Transport

Global Growth in Aviation



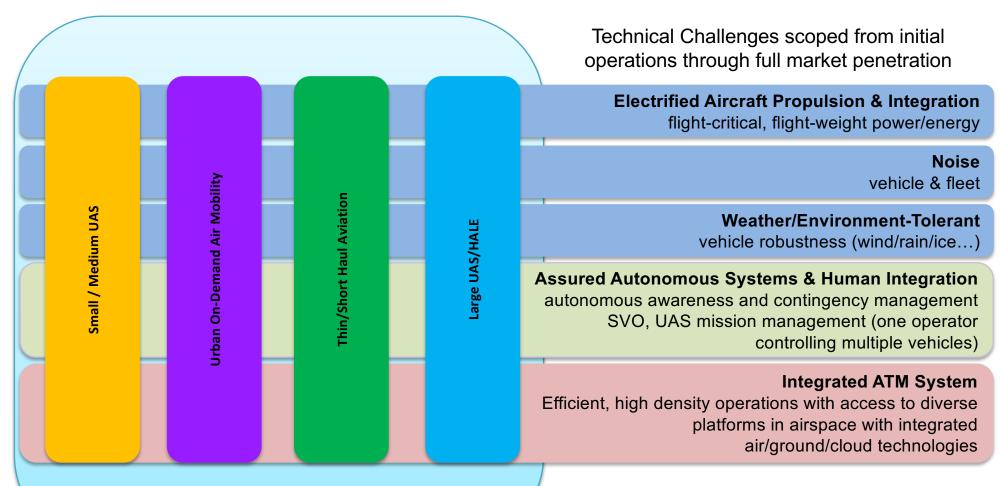


Emerging Markets - Integrated Challenges

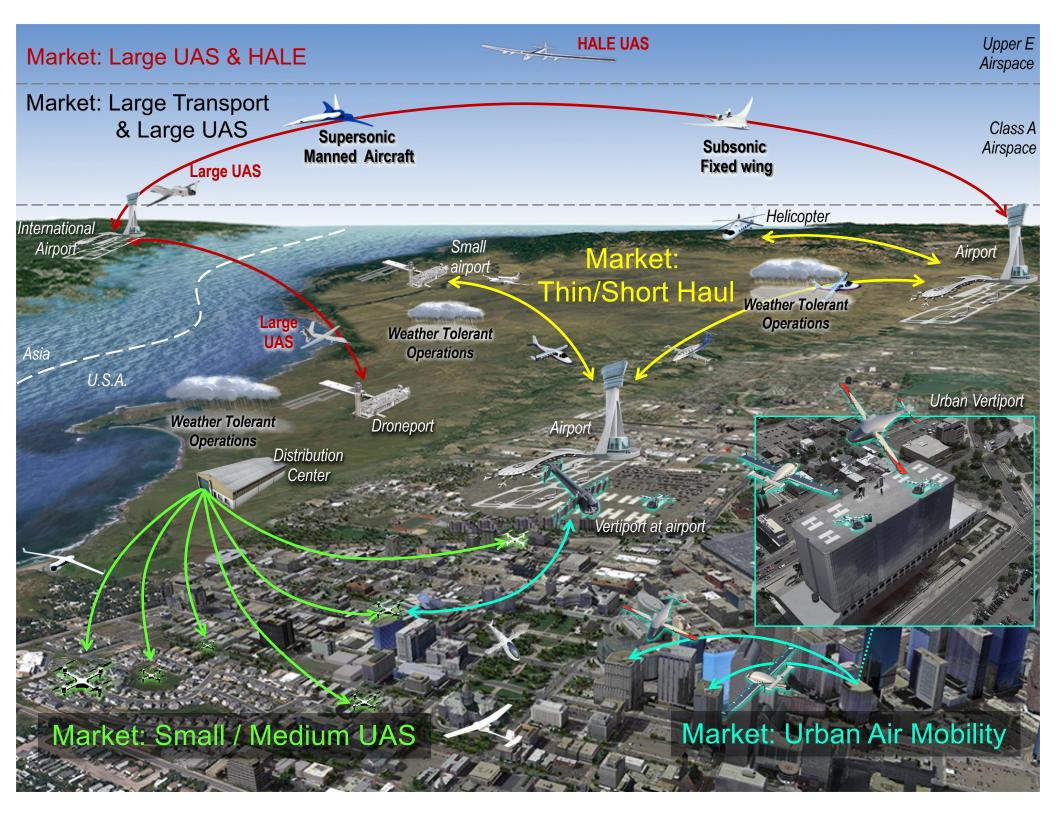
NASA ARMD Programs pivoting to address complex challenges



ARMD has developed a holistic understanding of the challenges for enabling the enormous potential of emerging aviation global market opportunities



Integrated Aviation System Challenges



NASA Aeronautics Strategic Implementation Plan

Continues to Guide NASA Aeronautics Investment





ARMD Research Programs & Strategic Thrusts



Airspace Operations & Safety

Projects

- Airspace Technology
 Demonstrations
- UAS Traffic
 Management
- System-Wide Safety
- ATM-X



Advanced Air Vehicles

Projects

- Advanced Air Transport Technology
- Advanced Composites
- Revolutionary Vertical Lift Technology
- Commercial Supersonic Technology
- Hypersonic Technology



Transformative Aeronautical Concepts

Projects

- Convergent Aeronautics Solutions
- Transformational Tools and Technologies
- University Innovation



ARMD Strategy https://www.nasa.gov/aeroresearch/strategy

NASA Aeronautics Vision for Aviation in the 21st Century



SEEDLING PROGRAM

MISSION PROGRAMS

FY 2020 Budget Request - Aeronautics

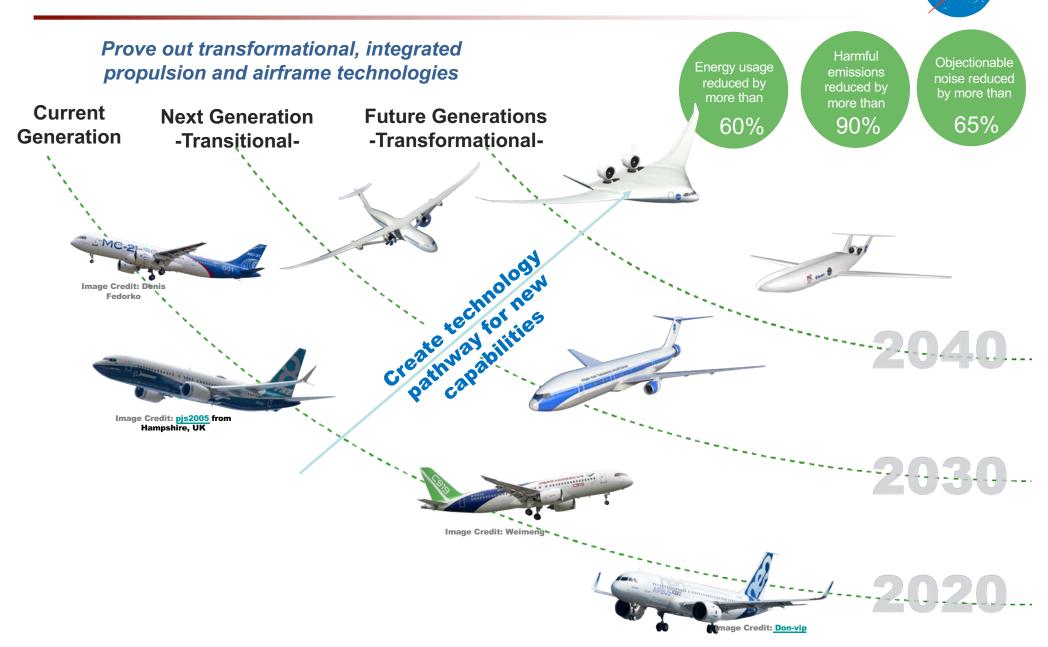


\$ Millions	FY 2018	FY 2019	FY 2020	FY 2021	FY 2022	FY 2023	FY 2024
Aeronautics	\$690.0	\$725.0	\$666.9	\$673.6	\$680.3	\$587.1	\$587.0
Airspace Operations and Safety	118.7		121.2	130.6	133.5	136.2	138.9
Advanced Air Vehicles	237.7		188.1	203.3	212.2	219.3	224.2
Integrated Aviation Systems	221.5		233.2	209.4	202.2	97.1	87.2
Transformative Aeronautics Concepts	112.2		124.4	130.3	132.3	134.6	136.7

FY 2018 reflects funding amounts specified in Public Law 115-41, Consolidated Appropriations Act, 2018, as adjusted by NASA's FY 2018 Operating Plan. FY 2019 reflects funding as enacted under Public Law 116-06, Consolidated Appropriations Act, 2019

Note: PBR FY20 and beyond budget decrease relative to FY19 is result of AETC transfer out of ARMD

Subsonic Transport Technology Strategy



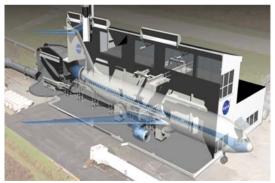
Enabling U.S. Leadership in Subsonic Transport Markets

- Suite of 5 key technologies coupled into transformative configurations will win the subsonic transport future
 - Light Weight, Very High Aspect Ratio Wings
 - Propulsion Airframe Integration, especially Boundary Layer Ingestion
 - Tailored Non-Circular Fuselage
 - Electrified Aircraft Propulsion
 - Small Core Turbine Engines
- ARMD is advancing these key technologies to create market opportunities



Boundary Layer Ingestion





NASA Electric Aircraft Testbed (NEAT) Facility

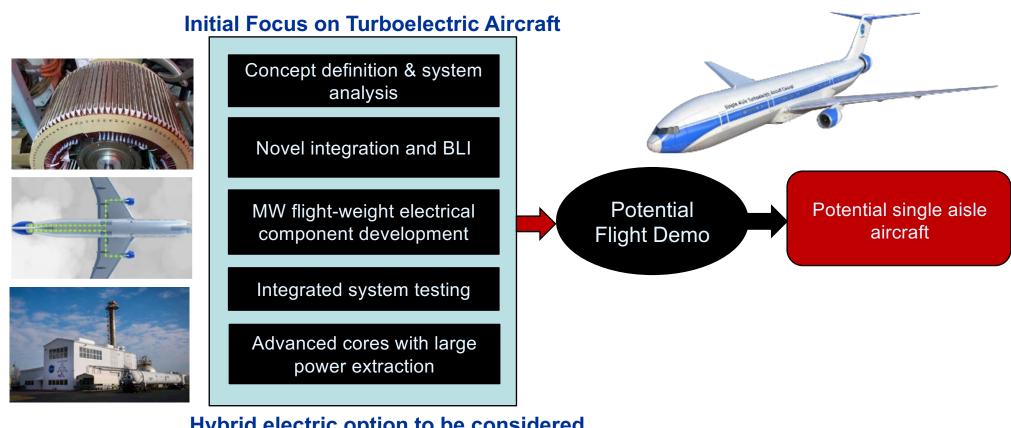


Very High Aspect Ratio Wing



Electrified Aircraft Propulsion Strategy for Single Aisle Aircraft





Hybrid electric option to be considered with advances in battery technology

MW-level High Efficiency/Power Density Electric Machines & Flight-weight Power System/Electronics

- Explore conventional & non-conventional topologies; Integrate novel thermal management; Demonstrate component maturation
- Develop/demonstrate powertrain systems and components; High voltage, MW power electronics, transmission, protection

Enabling Materials

- Insulators & conductors for high power & altitude components; Nanocomposite magnetic materials for targeted machines and drives

Integrated Subsystems

- Explore component interactions, power management, & fault management

Alternative Jet Fuel Combustion Research at NASA



TTT

- Fuel Sensitive Chemistry Models
- Fuel Sensitive simulation capability
 for Lean Blowout (LBO) and Ignition
- Improved Soot Models
- Testing with alternative fuels



Instantaneous temperature contours from LES of Referee Rig at near LBO condition

AATT

- Testing with alternative fuels (100% and blends)
- Flow reactor network for assessing fuel impacts over large range of geometries/conditions than practical with detailed CFD simulations



NASA Test Cell Ce-5 Stand 1 tested N+3 combustor with JetA and alternative fuel blend

NJFCP

- Provided experiments enabling development of chemistry models for conventional and alternative jet fuels
- Provided LBO and Ignition experiments with multiple fuels for development and validation of fuel sensitive CFD models

CST

- Fuel impacts on soot emissions
- Fuels optimized for flame stability and lower emissions

AATT-CGT: N+3 Low Emissions – Fuel Flexible Combustor Concept

Problem

Fuel flexible, Small Core, High OPR (50+) engines present greater challenges to combustor emissions and operability

Objective

Develop and demonstrate (to TRL 3) a low-emissions, fuel-flexible, smallcore compatible combustor architecture that reduces LTO NOx 80% below CAEP/6 and cruise NOx 80% below 2005 best-in-class.

Results

- Developed small-core lean-burn combustor technology for propulsion engines in efficient N+3 aircraft such as the D8 concept aircraft.
- Completed emissions testing of small core combustor in CE-5 covering LTO and cruise conditions.
- Completed testing using Jet-A and a 50/50 blend of Jet-A / Alternative fuel (Gevo Alcohol-to-Jet). Demonstrated equivalent combustor performance and the effect of fuel composition on nonvolatile particulate matter emissions.
- Demonstrated LTO NOx emissions >80% below CAEP6 in UTRC single-sector tests of small-core N+3 combustor.

Significance

Lean Burn combustor concept shows potential to meet/exceed NASA's Far Term LTO NOX goal. But further investigation of dynamic characteristics of the combustor concept are required to ensure good operability over the flight conditions

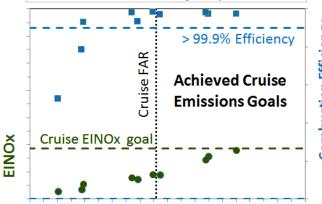
Emissions goals achievable with small core compatible combustor with Alternative Fuels



Summary of Emission Result

Flight Condition	<u>El NOx</u>			
Takeoff (100%)	5.01			
Climbout (85%)	3.81			
Approach (30%)	10.60			
Idle (7%)	4.40			
Dp/Foo	8.37 g NOx/kN			

[%] Below CAEP/6 Stringency: 88.2 % Below CAEP/8 Stringency: 86.6







ICAO/FAA Interaction and Technical Support

Landing/Takeoff Noise and Emissions Procedures for Supersonic Transport

Objective

 Analyze representative near-term commercial supersonic design space to inform CAEP regulators with LTO performance for use in Type Certification process.

Approach

- Use conceptual supersonic derivative mixed flow turbofan based on CFM56-7B and down-sized Boeing N+2 vehicle.
- Assess certification margins using conventional & advanced procedures for noise (WG-1).
- Assess certification margins using current Rich-Burn & advanced Lean-Burn combustor emissions for both conventional & advanced takeoff procedures (WG-3)
- Interact with industry for consensus on methods/assumptions

Status

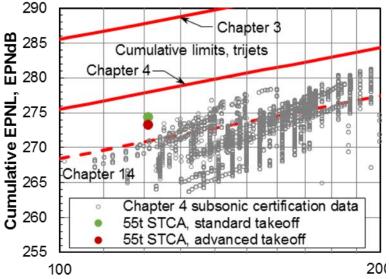
- Advanced takeoff procedures will be helpful in achieving Chapter 4 noise levels, but will require departures from subsonic standards/ reference procedures
- Existing subsonic CAEP/4 levels of LTO emissions appear achievable near-term, depending on regulation times-in-mode for take-off, climbout, approach, taxi.

Significance

NASA results are key to moving forward with CAEP gap analysis

STCA1: Range = 4000 n.mi. Mach = 1.4





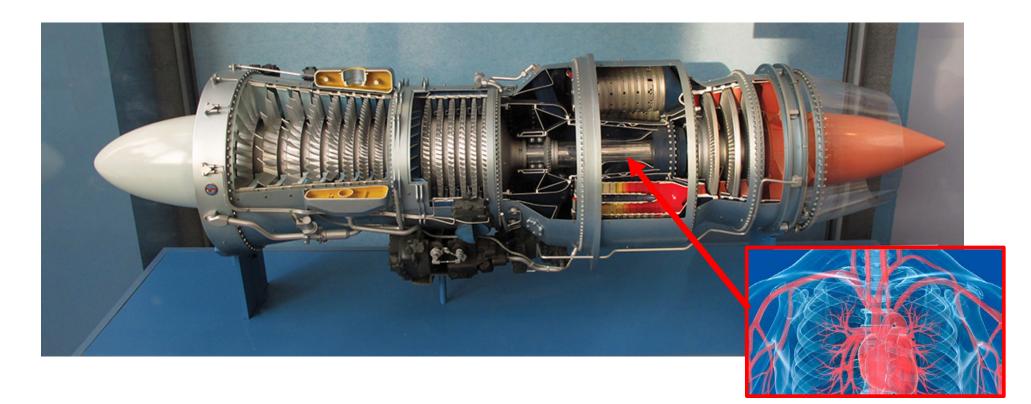
Maximum Takeoff Gross Weight, 1000lb

55t STCA EPNL predictions (with wing shielding) compared to Chapter 4 data.



Every engine has a heart





Loss of sight in the importance of the combustor is like forgetting about ones heart that supplies our entire body with the energy we need to get up every day. Not an option.