FAA CENTER OF EXCELLENCE FOR ALTERNATIVE JET FUELS & ENVIRONMENT

#### Noise Power Distance Re-evaluation Project 43

Lead investigator: Dimitri Mavris (PI), Chris Perullo (Co-I), Michelle Kirby (Co-I) Project manager: Hua (Bill) He (FAA)

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## **Project 43 Goals**



- Motivation
  - NPD method within AEDT was developed decades ago with little flexibility to account for airframe noise and speed effects
  - Away from airports and for different flight segments, assumptions become less robust
- Project Impact
  - Enhance the accuracy of AEDT through improved aircraft source noise prediction and modeling
  - Needed to support the evaluation and development of aircraft flight procedures that could reduce community noise
  - Facilitate the implementation of NextGen through improved characterization of the noise benefits it would deliver
- Objectives
  - Study representative fleet mixes and aircraft types
  - Validation against available measurement data
  - Investigate a method to effectively represent the fleet
  - Maintain compatibility with existing NPD (integrated modeling) approach

#### **ASCENT Project 43 Overview (Year 1&2)**



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• Objectives

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- Understand the sensitivity of including aircraft configuration changes and reference speed in NPDs on resulting noise contours for 50 – 400 PAX
- Provide physics-based recommendations on format of NPD + Configuration (NPD+C) curves for use in AEDT
- Maintain compatibility as much as possible with existing NPD approach



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#### **Summary of Key Findings**

- Examined six aircraft classes ranging from regional jet to large wide-body
- Found effect of flight velocity on source noise to be main source of difference
- Flap noise secondary contributor
- Major differences occur during approach
  - Engine noise near maximum power dominates during departure



	Grouping	Study	Parameters
	Baseline	0	Baseline NPD
		I.A	Include only speed
	Main Effects	I.B	Include only flaps/slats
		I.C	Include only gear
		II.A	Speed + Gear
	Cross Terms	II.B	Speed + Flaps
		II.C	Gear + Flaps
		II.D	Speed + Gear + Flaps



## **Various Options and approaches**



Integration Approach	Ву	Benefits	Challenges
Multi configuration NPDs (working w. mfgrs)	Eurocontrol	<ul> <li>From manufacturers.</li> <li>Considered to be well validated.</li> </ul>	<ul> <li>Only limited models so far.</li> <li>Challenges to cover fleet, esp. with out of production a/c models</li> </ul>
Fleet updated NPD+C directly from ANOPP	GT tried this	<ul> <li>The process is easy to understand</li> <li>Consistent method for generating NPD+C</li> </ul>	<ul> <li>Complex input parameters and delicate balance of the parameters</li> <li>Validation is still needed</li> <li>Large model library required</li> </ul>
NPD+C via correction functions based on ANOPP	GT (Proposed approach)	<ul> <li>Able to create NPD+C sets from simpler inputs (available within AEDT).</li> <li>No need to create ANOPP models for each a/c type.</li> </ul>	<ul> <li>Need to consider wide condition ranges/rank orders</li> <li>Validation of NPD+Cs</li> <li>Industry buy-in</li> </ul>

#### Motivates Simpler Implementation Approach

## **Upcoming Validation Work**

![](_page_5_Picture_1.jpeg)

- Task 1: Investigate impact of frequency content on standard NPD
  - How is spectral data used in AEDT?
  - ID what parameters to vary and how do they vary over time
  - What are the major drivers?
- Task 2: Investigate impact of frequency content along with NPD+C
  - To understand the current spectral class development process and effect when aircraft specific spectral data or even the power-setting/flap setting specific data are available.
  - Identify how to add multiple spectral data to an AEDT dep/app procedure, holding all other parameters constant for an aircraft Determine how to interpolate spectra
  - Leverage Volpe work and coordinate with aircraft manufactures to access data submitted for ANP
  - Conduct sensitivity studies using detailed spectral data available
  - Provide recommendations to the FAA on the results
- Task 3: Validate NPD+C Approach using airport noise monitoring data at a major US airport

# Task 1: Sensitivity Analysis of use of Spectral Data

![](_page_6_Picture_1.jpeg)

- Environmental impacts
  - Noise (SEL, LAmax, EPNL, PNLmax), emissions (NOx), and fuel consumption
- Weather parameters
  - Temperature, sea-level and station pressure, dew point, relative humidity, and wind speed (and cloudiness)
- Airports
  - SFO, ATL, DEN, and MEX
- Aircraft
  - CRJ900
  - B737-800 no winglets
  - B767-300ER
  - B777-200ER w/ GE engines
- Model all stage lengths with a 15,000 ft. cutoff altitude

#### **Weather Sources Example**

![](_page_7_Picture_1.jpeg)

• Weather sources

Iowa State University Environmental Mesonet (https://mesonet.agron.iastate.edu/request/download.phtml?network=GA\_AS OS)

- Weather data can be downloaded as CSV
- Daily weather data back to 1928 needs some processing

![](_page_7_Picture_6.jpeg)

# **Historical Weather Data**

![](_page_8_Picture_1.jpeg)

- Utilize historical weather to determine the bounds for each parameter
- Execute a DoE to determine noise metric sensitivity to spectral/weather data

![](_page_8_Figure_4.jpeg)

![](_page_8_Figure_5.jpeg)

![](_page_8_Figure_6.jpeg)

- Results will provide insight to the uncertainty in weather and noise propagation for Task 3
- Collaborate with PSU (Vic) to acquire 3D weather data from Spire Global for validation purposes

#### Task 2: Investigate Impact of Frequency Content on NPD+C

![](_page_9_Picture_1.jpeg)

- New students are getting up to speed on EDS/ANOPP and the prior analysis conducted with the multiple NPDs
  - Repeating prior analysis to ensure consistency
  - Investigating the spectral data sensitivity to changes in speed and configuration
  - Developing automation scripts
- Another portion of the team is resurrecting the AEDT code modifications made ~2+ years ago to handle multiple NPDs so as to gain insight on how to modify it to handle multiple spectral data
  - Challenge: that working version of AEDT is out of sync with the current public release version AEDT3c
  - This is doable, it will just take a lot of code modifications
- Ideally, GT would like to hand over the code modifications maintenance to the AEDT development team

### **Initial Spectral Sensitivity**

![](_page_10_Picture_1.jpeg)

- Approach spectral data is much more sensitive to flap and speed settings, which will require modeling in AEDT
- As expected, little variation on departure since noise is dominated by the engine, which will not require modeling in AEDT

![](_page_10_Figure_4.jpeg)

#### **Task 3: Validation**

![](_page_11_Picture_1.jpeg)

- Potential challenges in using measurement data for source noise validation
  - Uncertainties in aircraft state
    - Flap/slat setting
    - Thrust setting
    - Weight
    - Speed
  - Uncertainties in the atmospheric conditions
    - Humidity alone may cause a large variation in noise measurement
    - Wind speed/direction
  - Errors in noise propagation model
- How to overcome?
  - Use data from higher fidelity sources such as FOQA to reduce uncertainties in aircraft state (flap, thrust, weight etc.) and weather
  - Coordinate with the right individuals/companies to minimize the uncertainty

#### **SFO Noise Monitor Data**

![](_page_12_Picture_1.jpeg)

- GT's ASCENT Project 45 team had a telephone interview with SFO's Bert Ganoung, Manager of Aircraft Noise Abatement Office
- The interview was mainly about the history of NADP usages at SFO
- SFO has a very successful "Fly Quiet" program launched in 2002
- As part of the program, SFO has been operating 30+ noise monitors around the airport
- It uses an ANEEM system that utilizes the radar data to map a sound recording to a flight
- SFO is willing to share the noise and the radar data with GT

![](_page_12_Picture_8.jpeg)

https://www.flysfo.com/community/noiseabatement/fly-quiet

#### **Create A Fixed Point Profile Path Study Manually**

![](_page_13_Picture_1.jpeg)

![](_page_13_Figure_2.jpeg)

## **Fixed Point Profile Method**

![](_page_14_Picture_1.jpeg)

- The most straight forward way to model a FOQA flight is through AEDT's fixed point profile method
- Steps
  - Gather the entire flight data from FOQA
  - Filter to collect data at altitudes below 15,000ft AFE and remove taxing, maintaining key feature of trajectory
    - Automated script to select appropriate segment points to retain trajectory shape, thrust and speed trends, minimize sampling errors
  - Populate in SQL database
  - Create point ground track in AEDT GUI with lat-long data acquired from FOQA
  - Match created point profile to ground track when modeling in aircraft operations
- Comparison
  - Once fixed point profile outputs are obtained, compare with noise monitoring data for validation (ongoing)

# **Fixed Point Profile Modeling Example**

![](_page_15_Picture_1.jpeg)

#### Compare AEDT and original FOQA data to ensure reasonable agreement

![](_page_15_Figure_3.jpeg)

#### **Summary/Next Steps**

![](_page_16_Picture_1.jpeg)

Summary of current efforts under ASCENT 43 project are:

- 1. Spectral sensitivity
  - Initiated modeling plan for sensitivity tests and developing scripts to automate the process
  - Complete tests within the next few months
- 2. Spectral sensitivity with NPD+C
  - Working on modifications to AEDT source code to handle multiple spectral data sets
  - Once completed, the sensitivity analysis can begin
- 3. Validation with real world data
  - Gathering and modeling airline FOQA data in AEDT
  - Working with SFO on the noise monitoring data for the associated FOQA flights
  - Initiating comparison of real world data to AEDT standard profile

#### Acknowledgements

![](_page_17_Picture_1.jpeg)

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- Mike Doty, NASA Langley

#### **Publications**

- "Investigation of Aircraft Configuration and Speed on Traditional Noise-Power-Distance Curves" –NOISE-CON 2019
- A-21 meeting in DC (June 12 13) coordinate with industry and European efforts

#### **Participants**

- GT Research Staff:
  - Michelle Kirby, Chris Perullo, Tejas Puranik, Yongchang Li, Don Lim (now at Boeing)
- GT Students:
  - Ameya Behere, Seulki "Connor" Kim, Sarah Malak, Shilpa Ravoory, Andrew Van Zwieten, Max Fernandez